



SUPPLEMENTAL TYPE CERTIFICATE

10040586

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

AIRCRAFT STRUCTURES INTERNATIONAL Corp.

ENID WOODRING REGIONAL AIRPORT, HANGAR 31
RR 5, BOX 41B
ENID OK 73701
USA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Product TC Number : EASA.IM.A.226
TC Holder : CESSNA AIRCRAFT COMPANY
Model : CESSNA 208, CESSNA 208B
Original STC Number : FAA STC SA10840SC

Description of Design Change:

Installation of Jack Pads on Cessna 208 Series aircraft in accordance with FAA STC SA10840SC

EASA Certification Basis:

The Certification Basis for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency,

Date of issue: 11.07.2012


European Aviation Safety Agency
Manfred REICHEL
Project Certification Manager
General Aviation

Note:
The following numbers are listed on the certificate:
EASA current Project Number: 0010018271-001

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Associated Technical Documentation:

ASIC Master Data List ASIC-351 MDL, Rev. 5, dated March 23, 2010 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision).

Limitations:

None

Conditions:

Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product.

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Note:

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EASA current Project Number: 0010018271-001

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